

National Aeronautics and Space Administration



The Effects of Space Weather on Space Situational Awareness

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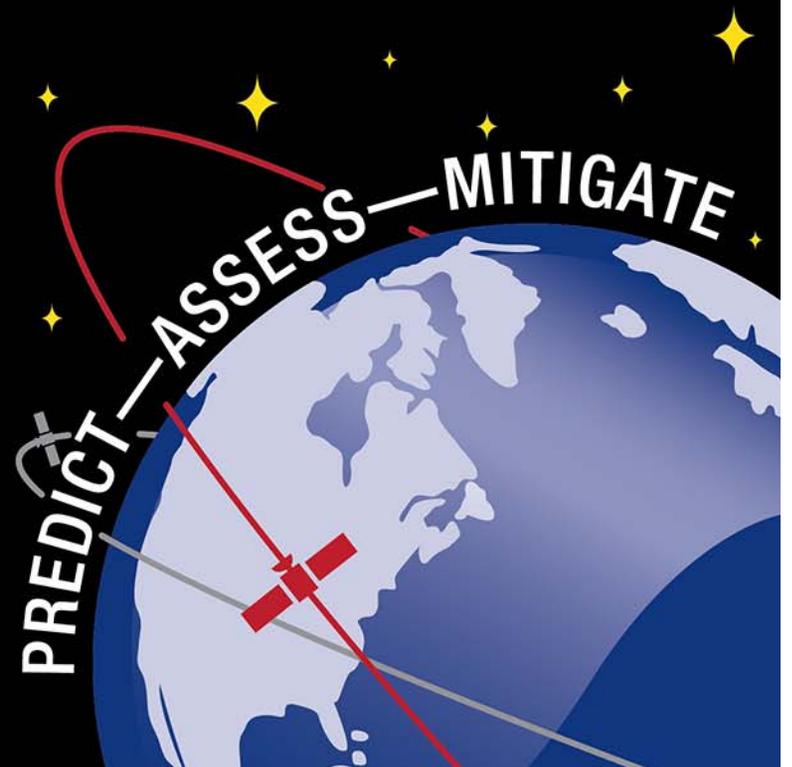
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Space Weather Enterprise Forum

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NASA ROBOTIC CARA

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Background: Space Weather Impacts on Satellites



- Particle bombardment
 - Electrical charging
 - Ionization events
- Satellite disorientation
- Communication loss
- Increased atmospheric drag
 - Satellite position
 - Covariance size
- Ionospheric effects
 - Incomplete/inadequate ionospheric correction will impact range performance of SSN ground-based radar tracking
 - Could impact OD

Important
to
Orbital
debris
profiling
and CA

Atmospheric Density Effects on SSA



- Historically, space surveillance networks have relied on two principal parameters to modify atmospheric models for solar activity variations.
 - F10.7: The solar 10.7 cm radio flux is indicative of the average level of radiation over the full solar disk
 - A_p : The geomagnetic index is strongly influenced by solar flares
- Changes in atmospheric drag will cause a satellite to penetrate a sensor's coverage volume at a time other than that expected.
 - Correlation of the object with its true identity might be delayed or not possible
 - Uncorrelated observations increase data processing requirements and can lead to less frequent updates of orbital elements and to "lost" satellites
- A decrease in the accuracy of satellite orbital parameters will also degrade the quality of conjunction assessments needed for collision avoidance decisions
- Unreliable predictions of atmospheric density will affect projections of debris populations for different orbital regimes

Lost Satellites



- A “lost satellite” is one for which reliable orbital elements are more than 30 days old
 - Satellites with orbits older than 5 days are typically placed on an attention list
 - Conjunction assessments conducted with data older than 5 days are of limited value
- The number of lost satellites typically grows during periods of high solar activity, due to difficulties in predicting neutral atmospheric density
 - The problem was well-documented during Solar Cycles 21 and 22
- In July 2012, approximately solar max, the lost list for the U.S. Space Surveillance Network reached a record high
 - Many factors contribute to lost list size, but atmospheric density mis-modeling plays a role

Other SSA Considerations



- The satellite catalog today consists of a much greater number of very small objects with moderate or high area-to-mass ratios. Orbit maintenance for such objects is more challenging, in part, because only one or a few sensors can track them
- Increased orbital decay rates can result in more objects passing through a given orbital regime, increasing the need for conjunction assessments
 - The International Space Station has experienced a marked increase in close approaches as more objects fall through its orbital regime
 - Prediction of debris densities, and therefore prediction of future debris environments and shielding requirements, also affected adversely
- Changes in atmospheric ionization can adversely affect the performance of radar sensors, the primary means of monitoring objects in low Earth orbits
 - Range errors can increase
 - Low elevation passes might need to be restricted, leading to shorter observation times or missed opportunities

Potential SSA Countermeasures



- Increase in tracking frequency for many satellites
 - Places greater loads on both sensors and processing centers
- Increase in frequency of satellite orbital element updates
 - Orbital elements are often not updated after each new observation
- Increase in the use of full differential correction in orbital element updates
 - Often only partial differential corrections are performed
- Increase in the frequency and quality of updates of solar indices
- Improvement of atmospheric models
 - High accuracy satellite drag model (HASDM) in development in U.S. since 2001; improved HASDM to be operational in 2013
 - However, still empirical rather than physics-based model; will be limited in predictive capabilities
- Modification of collision avoidance processes and procedures to take into account the increased uncertainty (covariance) in conjunction assessments
 - Nonetheless, will increase both type 1 and type 2 errors

NASA Robotic Conjunction Assessment Risk Assessment Process



Air Force
(JSpOC)

- JSpOC Orbital Safety Analyst**
- Generate daily close approach predictions - provide results to Risk Assessment Team
 - Interface with Space Surveillance Network - request increase tracking data collects
 - Provide state and state uncertainty information for primary and secondary objects at TCA
 - Provide Miss Distance Summaries



CA Predicts
OD Quality Assessment
Event Discussion
Data Products

NASA
Robotic
CARA

- NASA/Robotic Risk Assessment Team**
- Perform Collision Risk Assessment Analysis
 - Quantify Collision Risk
 - Compute Collision Probability
 - Perform probability Sensitivity Analysis
 - Interface with JSpOC CA team to ensure data product delivery and tracking requests
 - Assist Mission Operations with avoidance maneuver strategy
 - Provide recommendations to Mission Operations Team and Mission Management



- Mission Management**
- Make final maneuver decision

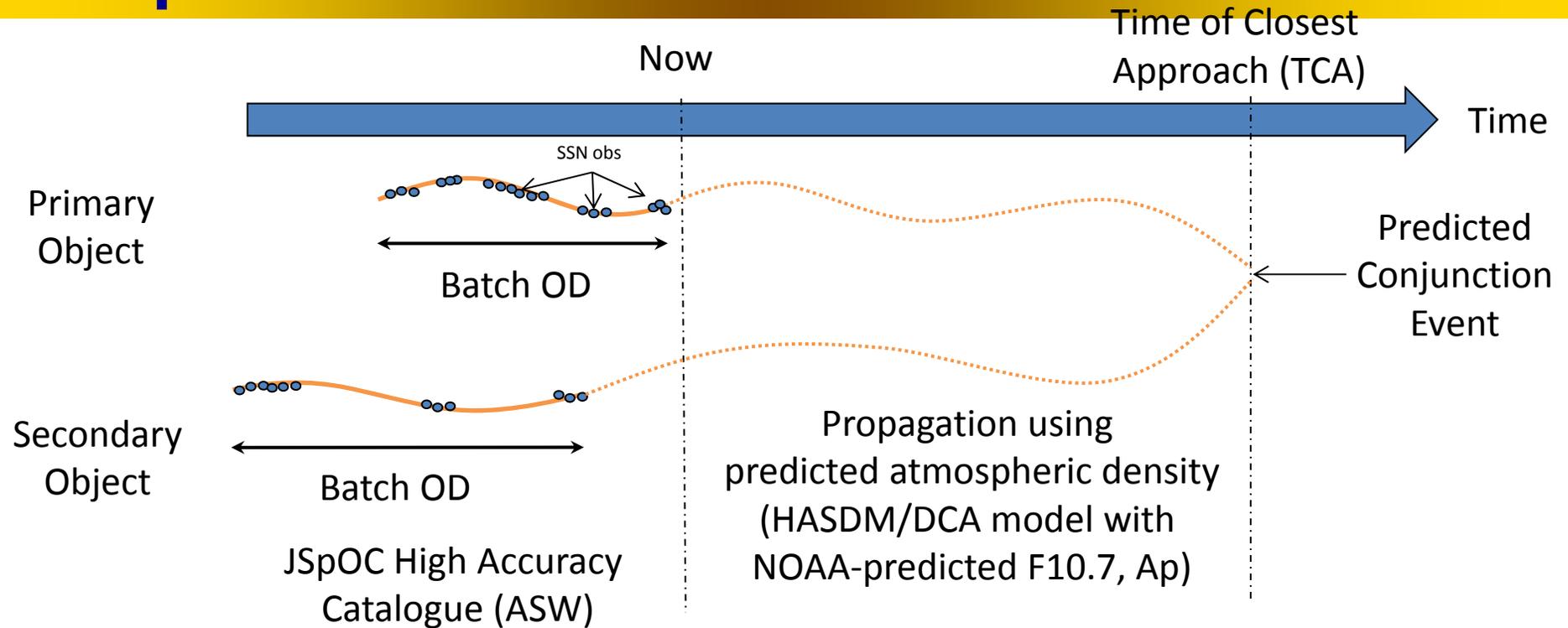


- Mission Operations Team**
- Avoidance maneuver planning & maneuver execution
 - Provide ephemeris (state & covariance) to Risk Assessment Team for evaluation
 - Provide recommendation to Mission Management



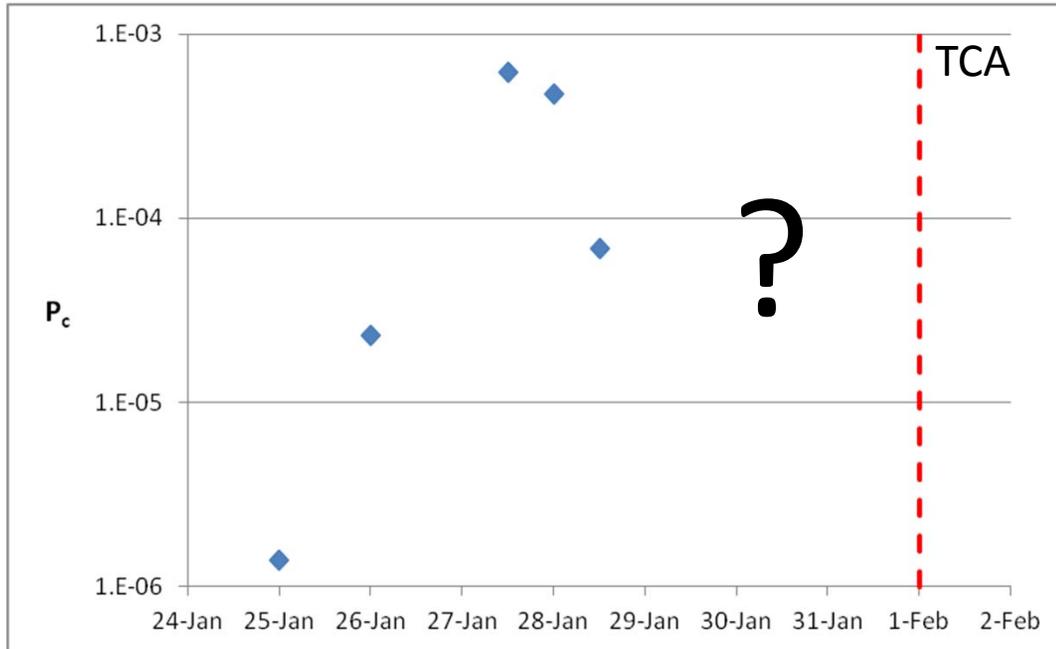
End User
(O/O)

Conjunction Assessment: JSpOC Process and Products



- Orbital Conjunction Message (OCM) provided for each screening:
 - Includes both objects' state vector and position covariance at TCA
 - Allows computation of probability of collision (P_c)
 - Higher drag in the *past* (during OD) leads to larger covariance size in *future* (TCA)

Space Weather and Conjunction Assessment: A Notional Event



25 Jan: first identification of possible conjunction on 1 Feb

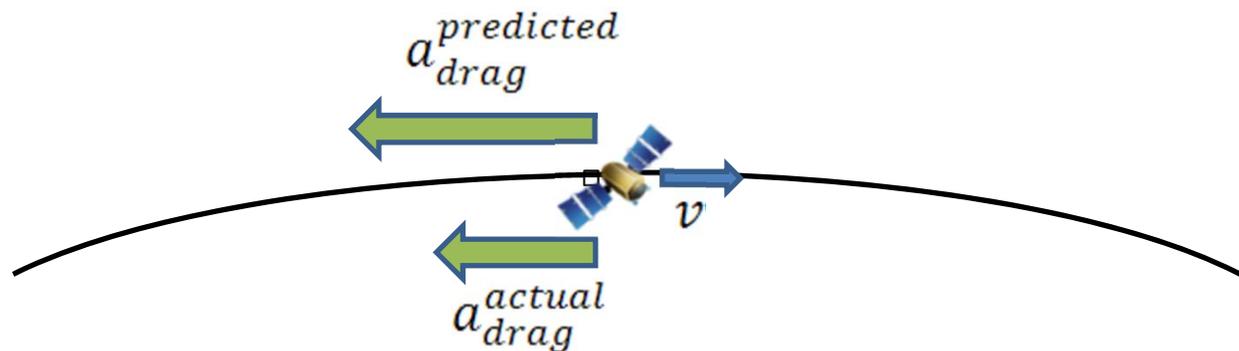
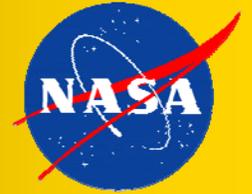
27-28 Jan: P_c first increases to level of concern before starting to fall (looking safer)

29 Jan: Alert of a Coronal Mass Ejection (CME) heading for Earth on 31 Jan

Spacecraft O/O wants to know if (and how) CME will impact conjunction event

- Does the new space weather prediction make this event safer or riskier?
- Might performing a maneuver make the conjunction event worse?

Background: Atmospheric Drag and Predicted Satellite Position



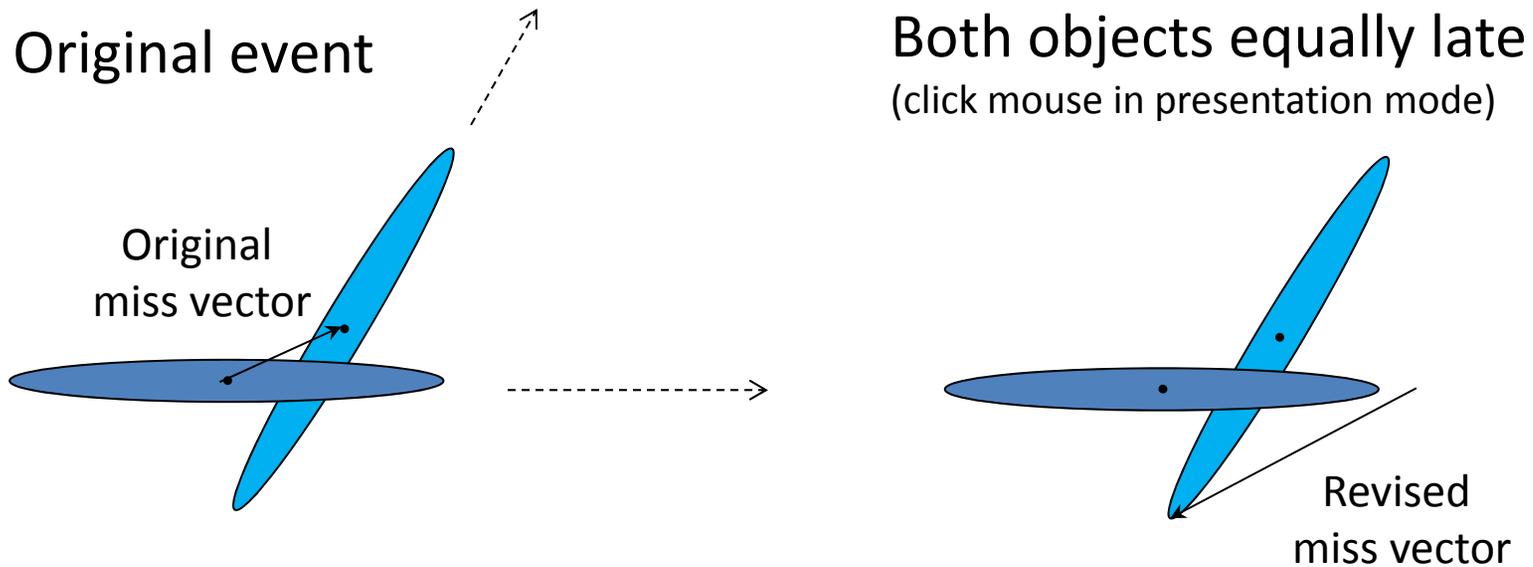
$$a_{drag}^{actual} \neq a_{drag}^{predicted}$$

- Satellite will be at a different position if
 - Uncertainty in predicted atmospheric density not currently incorporated into propagation results
 - Uncertainty in ballistic coefficient incorporated in covariance
- Drag acceleration ~counter to satellite velocity
 - Change to drag primarily results in offset in along-track position
 - Equivalently can be represented as an early/late offset time

Qualitative Impact of Time Offsets



"Non-head-on" event with different offset times:



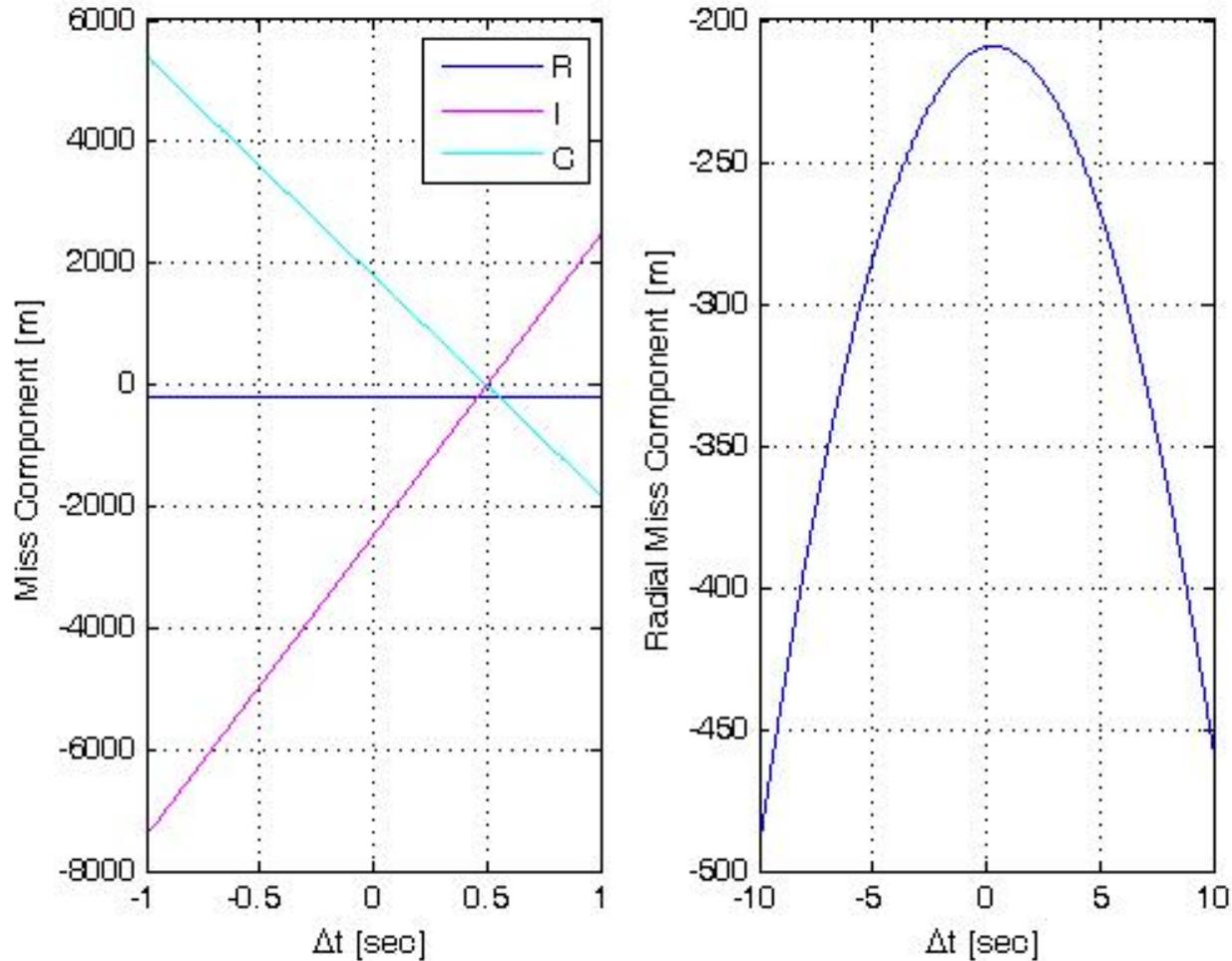
Results: TCA shifts, and conjunction event looks different at new TCA

How to Enhance Risk Assessment



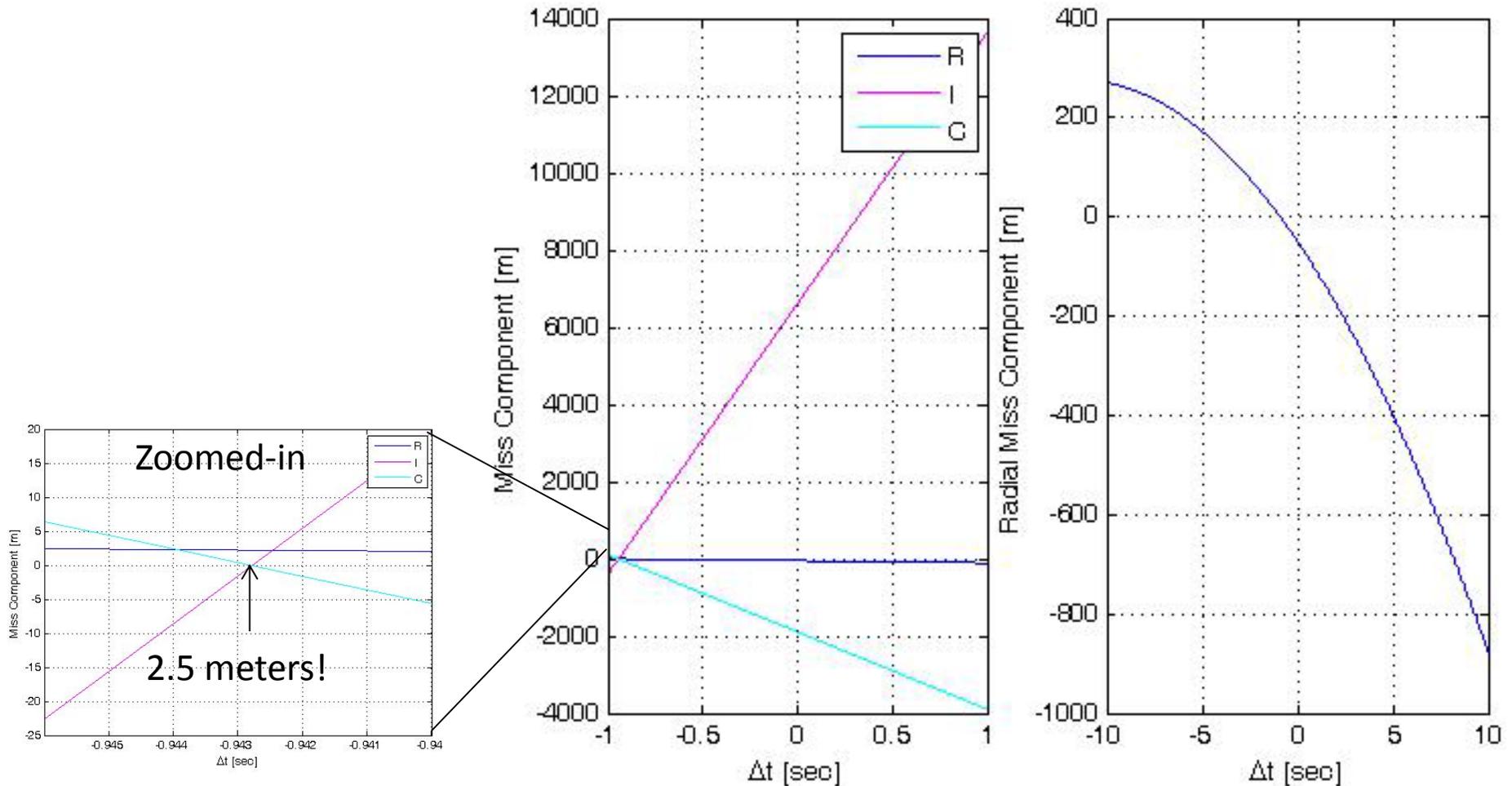
- Cannot at present determine actual effect of CME on atmospheric density
 - Atmospheric density will increase, and thus drag will increase; but cannot say by how much
- However, one can explore drag increases to determine if conjunction likely to become riskier
 - Increase in drag maps to time offset in objects' closest approach
 - Can profile situation for a variety of delta-t values and compute conjunction parameters
 - Can determine increased drag situation presents possible riskier scenarios than the nominal solution

Quantitative Impact of Time Offsets: No increase in risk



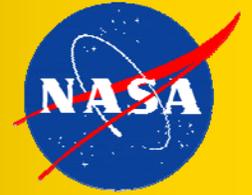
No risky scenario in negative half-plane

Quantitative Impact of Time Offsets: Potential Increase in Risk



There exists a negative-half-plane time offset that is extremely dangerous!

More Robust Approaches: Space Weather Predictions



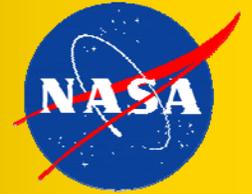
- Because satellite maneuver commit times occur ~2-3 days before TCA, conjunction risk is always determined using predicted data
- Non-trivial space weather predictions thus required
 - Three-day forward-predict of F10 and Ap available from USAF/NOAA
 - Predictions not particularly accurate
 - Propagations longer than three days lack full space weather predictions
- Present atmospheric model (J70MOD) is empirical model
 - Upgrading to JBH09 in next month—should bring 15-25% improvement depending on orbit altitude and propagation time
 - Will use eleven solar indices rather than two
 - Still an empirical model, however

More Robust Approaches: Physics-based Atmospheric Models



- Empirical models quite effective at modeling atmospheric density in the past
 - Useful for orbit determination
- Such models weaker in prediction, due to their nature
- Physics-based modeling a promising paradigm for prediction use, especially beyond 72 hours
- Also better suited for trying to model effects of discrete space weather events, such as coronal mass ejections (CMEs)
- However, probably 5-10 years before such a model developed, vetted, and available to operations

CARA Way Forward



- Deploy heuristic tool that determines whether increases in drag increase or decrease the risk of a particular conjunction event
 - Consider whether some maneuver planning rules can be developed for cases in which risk increased by CME &c.
- Await completion/deployment of more robust atmospheric models and predictive space weather indices
 - Bounds on Δt would be helpful, even though incomplete solution
 - General predictive improvements (e.g., longer than 72 hours) would be implemented by the JSpOC; probably no direct CARA implementation
 - Density effects of CMEs could be modeled directly by CARA for individual high-interest events